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# Lateral Dynamics and Stability Analysis of a Gas Compressor Supported by Hybrid and Active Lubricated Multirecess Journal Bearing

*Fluid film forces are generated in multirecess journal bearings by two types of lubrication mechanism: the hydrostatic lubrication in the bearing recesses and hydrodynamic lubrication in the bearing lands, when operating in rotation. The combination of both lubrication mechanism leads to hybrid journal bearings (HJB). When part of hydrostatic pressure is also dynamically modified by means of hydraulic control systems, one refers to the active lubrication, resulting in the active hybrid journal bearing (AHJB). The AHJB is mathematically modeled based on the Reynolds' equations in land regions and a set of modified continuity equations for the lubricant flow in the bearing recesses, coupled to equations describing pressure and flow in the servovalves. The solution of such a set of equations allows the determination of fluid film stiffness and damping coefficients of the hybrid and/or active lubricated bearing. Such coefficients are a function of design and operational parameters, characterized by the Sommerfeld number as well by the gains of the feedback control system. The main contribution of the present theoretical work is to analyze the stability characteristics of a flexible rotor-bearing system by using passively and actively lubricated hybrid journal bearing. The dynamic of a flexible rotor is modeled by using finite element method and, after coupling the bearing dynamic coefficients to the rotor model, the feedback control law is defined and a suitable set of control gains is calculated for the active lubrication, leading to a rotor with a wider and more safety operational frequency range.*

**Keywords:** Hybrid bearing, active lubrication, vibration control, stability

## Introduction

Aiming for the vibration reduction in rotating machines to safety levels, new mechanisms for dissipating vibration energy have been developed, like seal dampers (Vance and Li, 1996), squeeze-film dampers (San Andrés and Lubell, 1998), hybrid squeeze-film dampers (El-Shafei and Halthout, 1995), hydraulic active chamber systems (Ulbrich and Althaus, 1989; Althaus *et al.*, 1993; Santos, 1995), variable impedance hydrodynamic journal bearings (Goodwin *et al.*, 1989), actively lubricated tilting-pad bearings (Santos, 1994; Santos and Russo, 1998; Santos and Nicoletti, 1999; Santos and Scalabrin, 2003), active-controlled hydrostatic bearings (Bently *et al.*, 2000), magnetized journal bearings lubricated with Ferro fluids (Osman *et al.*, 2001), and actively lubricated hybrid multirecess journal bearings (Santos and Watanabe, 2003 and 2004).

Recent theoretical and experimental investigations related to active lubrication have been shown the feasibility of attenuating rotor vibrations in test rigs with rigid rotors (Santos, 1994; Santos and Russo, 1998; Santos and Nicoletti, 2001; Santos and Scalabrin, 2003; Santos *et al.*, 2001). The use of active lubrication in tilting-pad journal bearings (TPJB) has the strong advantage of resulting in bearings with negligible fluid film cross-coupling effects between orthogonal directions. However, this kind of active strategy can also be applied to hydrostatic or hybrid bearings (Bently *et al.*, 2000; Santos and Watanabe, 2003 and 2004), and to multi-lobed bearings (Santos *et al.*, 2001).

The main contribution of the present theoretical work is to analyze the feasibility of applying multirecess journal bearings under hybrid and active lubrication regimes to a flexible rotor

(industrial gas compressor) with the aim of attenuating rotor lateral vibrations. In the hybrid lubrication case, the hydrostatic pressure in all four recesses is statically modified, allowing an adjustment of the bearing stiffness. In the active lubrication case, the pressure and flow into opposed pair of recesses are dynamically modified with help of servo-hydraulic control. It results in a significant modification of fluid film forces and, consequently, in an active modification of bearing stiffness and damping coefficients. The mathematical model of hybrid journal bearings presented by Ghosh *et al.* (1989) and Guha *et al.* (1989) is extended by including the dynamics of servo control system, as presented by Santos and Russo (1998). A multirecess hybrid journal bearing with active lubrication is termed active hybrid journal bearing (AHJB). The AHJB static and dynamic coefficients are numerically calculated by solving the complete set of continuity equation in recesses regions and Reynolds' Equation in land regions, as a function of Sommerfeld number, perturbation frequency, servovalve dynamic characteristics and feedback control gains. These bearing coefficients, under hybrid and active lubrication regimes, are employed to analyze stability and feasibility of lateral vibration reduction in a gas compressor. The lateral dynamics of the gas compressor is modeled by using finite elements technique. By applying multirecess journal bearings under hybrid and active lubrication regimes to flexible rotor a significant reduction of lateral vibrations at critical speeds and a wider and more safety operational range are achieved.

## Nomenclature

$a$  = circumferential land width,  $m$

$B_{\epsilon\epsilon}, B_{\phi\phi}, B_{xx}, B_{yy}$  = direct damping coefficients,  $Ns/m$

$B_{\epsilon\phi}, B_{\phi\epsilon}, B_{xy}, B_{yx}$  = cross coupling damping coefficients,  $Ns/m$

$b$  = axial land width,  $m$

$c$  = radial clearance,  $m$

$D$  = bearing diameter,  $m$

$d$  = journal diameter,  $m$   
 $e$  = journal eccentricity,  $m$   
 $f_r$  = rotation frequency,  $Hz$   
 $g_{1X}, g_{2X}, g_{1Y}, g_{2Y}$  = direct gain coefficients,  $V/m$   
 $g_{3X}, g_{4X}, g_{3Y}, g_{4Y}$  = cross-coupling gain coefficients,  $Vs/m$   
 $h$  = fluid film thickness,  $m$   
 $\bar{h}_j^{(n)}$  = film thickness between  $n$ -th and  $(n+1)$ -th recesses  
 $\bar{h}_r$  = film thickness vector at the center recesses  
 $K_{PQ}$  = servovalve flow-pressure coefficient,  $m^3/sPa$   
 $K_V$  = servovalve gain,  $m^3/sV$   
 $K_{\varepsilon\varepsilon}, K_{\varphi\varphi}, K_{XX}, K_{YY}$  = direct stiffness coefficients,  $N/m$   
 $K_{\varepsilon\varphi}, K_{\varphi\varepsilon}, K_{XY}, K_{YX}$  = cross coupling stiffness coefficients,  $N/m$   
 $L$  = bearing length,  $m$   
 $L_e$  = effective recess length ( $L_e = L - a$ ),  $m$   
 $l_a$  = axial recess length,  $m$   
 $l_b$  = circumferential recess length,  $m$   
 $P$  = fluid film pressure,  $Pa$   
 $P_L$  = servovalve load pressure,  $Pa$   
 $P_r$  = bearing recess pressure,  $Pa$   
 $P_s$  = servovalve supply pressure,  $Pa$   
 $P_{s0}$  = bearing steady state supply pressure,  $Pa$   
 $\bar{p}_r$  = recess pressure vector  
 $\bar{Q}_r$  = recess flow matrix  
 $\bar{q}_s$  = fluid flow vector injected into the recesses  
 $R$  = bearing radius,  $m$   
 $r$  = journal radius,  $m$   
 $T$  = coordinate transformation matrix  
 $t$  = time,  $s$   
 $t_r$  = recess depth,  $m$   
 $U$  = servovalve input voltage,  $V$   
 $\bar{V}_e$  = effective recess volume matrix  
 $XYZ$  = inertial coordinate system  
 $xyz$  = local coordinate system

#### Greek Symbols

$\Delta e$  = journal dynamic displacement,  $m$   
 $\Delta \dot{e}$  = journal velocity,  $m/s$   
 $\Delta P_s^{(n)}$  = dynamic supply pressure in  $n$ -th recess,  $Pa$   
 $\delta_c$  = capillary parameter, dimensionless  
 $\varepsilon$  = journal eccentricity ratio, dimensionless  
 $\zeta_v$  = servovalve damping factor  
 $\Theta$  = recess squeeze parameter, dimensionless  
 $\theta$  = angular coordinate,  $rad$   
 $\kappa$  = fluid compressibility parameter,  $Pa^{-1}$   
 $\Lambda$  = angular velocity parameter, dimensionless  
 $\mu$  = fluid absolute or dynamic viscosity,  $Ns/m^2$   
 $\Pi$  = recess compressibility parameter, dimensionless  
 $\rho$  = fluid volumetric density,  $kg/m^3$   
 $\tau$  = time, dimensionless  
 $\Phi$  = recess velocity parameter, dimensionless  
 $\varphi$  = attitude angle,  $rad$   
 $\Psi$  = squeeze or perturbation, dimensionless  
 $\Omega$  = journal angular velocity,  $rad/s$   
 $\omega$  = perturbation frequency,  $rad/s$   
 $\omega_v$  = servovalve eigenfrequency,  $rad/s$

#### Subscripts

$0$  relative to static or steady state condition  
 $1,2$  relative to dynamic or perturbed conditions  
 $l$  relative to land region  
 $op$  relative to operation point  
 $r$  relative to recess

$s$  relative to supply conditions  
 $v$  relative to servovalve  
 $X, Y$  relative to  $X$  and  $Y$  directions, respectively  
 $\varepsilon, \varphi$  relative to  $\varepsilon$  and  $\varphi$  directions, respectively  
**Superscripts**  
 $(n)$  relative to  $n$ -th recess  
 $w^*$  relative to amplitude of generic parameter  $w$   
 $\bar{w}$  relative to dimensionless form of generic parameter  $w$

### Active Hybrid Journal Bearing - AHJB

The AHJB under investigation has four recesses, aligned in pairs in the horizontal ( $Y$ ) and vertical ( $X$ ) directions and numbered as shown in Fig.1a. The conventional passive bearing operation is warranted by fluid injection into the recesses through capillary restrictors, at a constant pressure supply  $P_{s0}$ . Additionally, for active dynamic control of the fluid pressure and compensation of cross-coupling effects, the flow is injected into the opposed bearing recesses pairs, each one of them are connected to servo control systems. Electro-hydraulic servovalves, journal displacement and/or velocity transducers and PD (proportional-derivative) feedback controllers, constitute the servo control system. The servovalves are controlled by electrical voltage signals,  $U_x$  and  $U_y$ , generated by the combination of journal dynamic displacement and velocity measurement signals in  $X$  and  $Y$  directions. The main geometric characteristics of a four recesses hybrid journal bearing, nomenclature and coordinate systems are represented in Fig. 1.

### AHJB Mathematical Modeling

The mathematical model of the HJB is obtained basically by using the Reynolds' equation in the land surfaces and the flow continuity equation in the bearing recesses, as presented by Santos and Watanabe (2003). The modeling technique used is the small perturbation method applied to the journal steady state equilibrium position (Lund and Thomsen, 1978, and by Ghosh et al., 1989). Within the bearing radial clearance  $c$ , the journal static eccentricity and attitude angle are defined by  $e_0$  and  $\varphi_0$ , respectively, and the journal dynamic perturbations are defined by the real part of the harmonic functions given by  $e_j^* e^{i\omega t}$  and  $\varphi_j^* e^{i\omega t}$ , where  $e_j^*$  and  $\varphi_j^*$  are the perturbations amplitudes. The resulting dimensionless fluid film thickness  $\bar{h} = h/c$  is defined as a function of the dimensionless eccentricity ratio  $\varepsilon = e/c$ , the attitude angle  $\varphi_j^*$ , the angular coordinate  $\theta = x/R$ , and the dimensionless time  $\tau = \omega t$ .

$$\begin{cases} \bar{h} = \bar{h}_0 + \varepsilon_j^* e^{i\tau} \cos \theta + \varepsilon_0 \varphi_j^* e^{i\tau} \sin \theta \\ \bar{h}_0 = 1 + \varepsilon_0 \cos \theta \end{cases} \quad (1)$$

Considering the small perturbation characteristics, the dimensionless fluid film pressure  $\bar{P} = P/P_{s0}$  may be described similarly to dimensionless film thickness  $\bar{h}$ , defined Eq.(1), as follows

$$\bar{P} = \bar{P}_0 + \varepsilon_j^* e^{i\tau} \bar{P}_1 + \varepsilon_0 \varphi_j^* e^{i\tau} \bar{P}_2 \quad (2)$$

where, the index  $0$  is relative to steady state condition and the indexes  $1$  and  $2$  are relative to dynamic or perturbed conditions.

The fluid film behavior in the land surfaces of a finite bearing is described by the three dimensionless Reynolds' equations, given in Eq.(3-5), deduced by using the Navier-Stokes, continuity equations

and Eq.(1-2), considering a isoviscous incompressible fluid, in laminar flow and including the hydrodynamic and squeeze effects.

$$\frac{\partial}{\partial \theta} \left( \bar{h}_0^3 \frac{\partial \bar{P}_0}{\partial \theta} \right) + \left( \frac{D}{L} \right)^2 \frac{\partial}{\partial \bar{z}} \left( \bar{h}_0^3 \frac{\partial \bar{P}_0}{\partial \bar{z}} \right) = \Lambda \frac{\partial \bar{h}_0}{\partial \theta} \quad (3)$$

$$\left. \begin{aligned} \frac{\partial}{\partial \theta} \left( \bar{h}_0^3 \frac{\partial \bar{P}_1}{\partial \theta} \right) + \frac{\partial}{\partial \theta} \left( 3\bar{h}_0^2 \cos \theta \frac{\partial \bar{P}_0}{\partial \theta} \right) + \left( \frac{D}{L} \right)^2 \frac{\partial}{\partial \bar{z}} \left( \bar{h}_0^3 \frac{\partial \bar{P}_1}{\partial \bar{z}} \right) + \\ + \left( \frac{D}{L} \right)^2 \frac{\partial}{\partial \bar{z}} \left( 3\bar{h}_0^2 \cos \theta \frac{\partial \bar{P}_0}{\partial \bar{z}} \right) = -\Lambda \sin \theta + i\Psi \cos \theta \end{aligned} \right\} \quad (4)$$

$$\left. \begin{aligned} \frac{\partial}{\partial \theta} \left( \bar{h}_0^3 \frac{\partial \bar{P}_2}{\partial \theta} \right) + \frac{\partial}{\partial \theta} \left( 3\bar{h}_0^2 \cos \theta \frac{\partial \bar{P}_0}{\partial \theta} \right) + \left( \frac{D}{L} \right)^2 \frac{\partial}{\partial \bar{z}} \left( \bar{h}_0^3 \frac{\partial \bar{P}_2}{\partial \bar{z}} \right) + \\ + \left( \frac{D}{L} \right)^2 \frac{\partial}{\partial \bar{z}} \left( 3\bar{h}_0^2 \cos \theta \frac{\partial \bar{P}_0}{\partial \bar{z}} \right) = \Lambda \cos \theta + i\Psi \sin \theta \end{aligned} \right\} \quad (5)$$

where,

$$\bar{z} = 2z/L$$

$$\Lambda = 6\mu\Omega/P_{s0} (c/R)^2$$

$$\Psi = 12\mu\omega/P_{s0} (c/R)^2$$

Equations (3-5) are solved by using the *finite difference* method with the boundary condition listed below for  $k=0$  (steady state condition) and for  $k=1,2$  (dynamic conditions)

(a)  $\bar{P}_0(\theta, \bar{z}) = 0$ , in the regions of cavitations (land surfaces)

(b)  $\bar{P}_k = \bar{P}_r^{(n)}$ , fluid pressure at the  $n$ -th recess

(c)  $\bar{P}_k(\theta, +l) = \bar{P}_k(\theta, -l) = 0$ , for symmetry

(d)  $\bar{P}_k(\theta, \bar{z}) = \bar{P}_k(\theta + 2\pi, \bar{z})$ , for periodicity

(e)  $\frac{\partial \bar{P}_k}{\partial \bar{z}}(\theta, 0) = 0$ , for symmetry

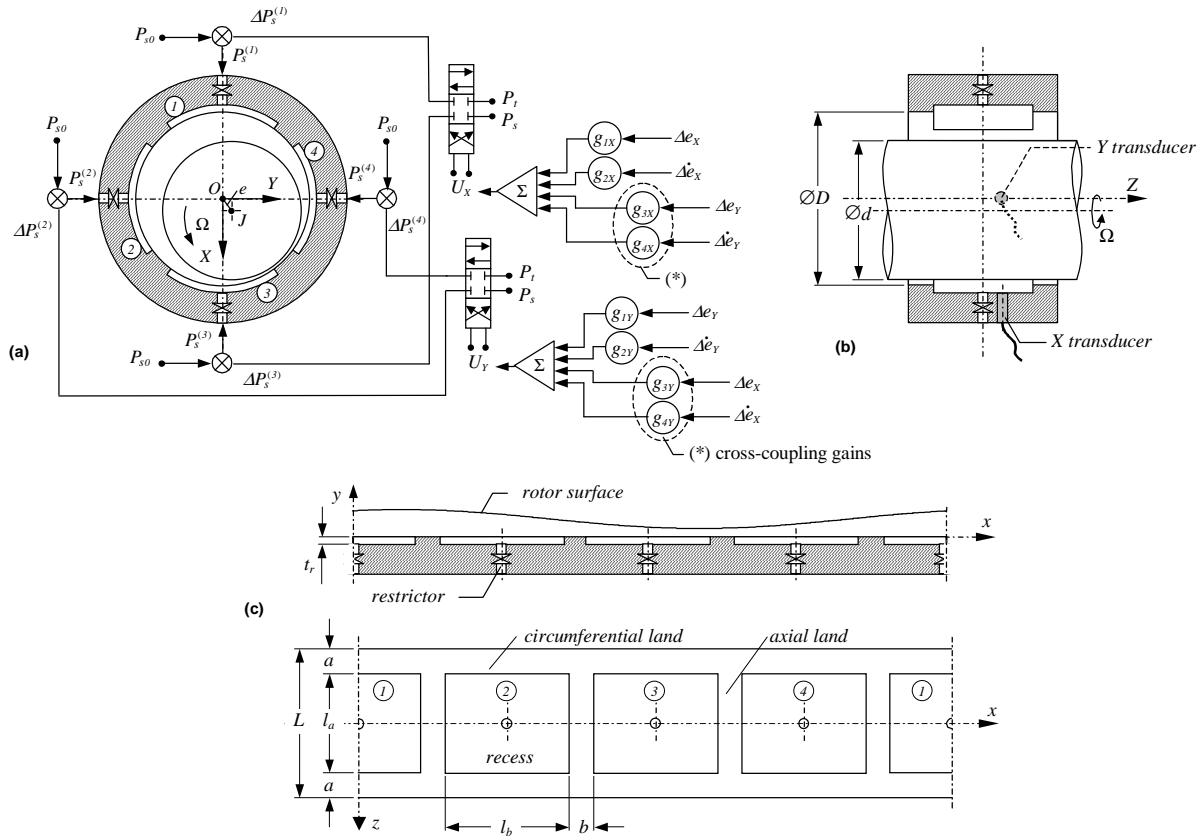


Figure 1. Active Hybrid Journal Bearing – geometric characteristics and operational principle.

The fluid pressures in the recesses are obtained by applying the continuity equation, considering fluid hydrodynamic, hydrostatic, squeeze and compressibility effects. The recess fluid pressures  $\bar{P}_r^{(n)}$ , in the  $n$ -th recess, is defined similarly to the fluid film pressure  $\bar{P}$  as follows

$$\bar{P}_r^{(n)} = \bar{P}_{r0}^{(n)} + \varepsilon_1 \bar{P}_{r1}^{(n)} e^{i\tau} + \varepsilon_2 \bar{P}_{r2}^{(n)} e^{i\tau} \quad (6)$$

Defining the dimensionless flow as follows,  $\bar{Q} = 12\mu Q/c^3 P_{s0}$ , and applying the continuity equation to all recesses, results to the following vector equation

$$\bar{q}_s = \bar{Q}_r \bar{p}_r + \Phi \Delta \bar{h} + \Theta \frac{\partial \bar{h}_r}{\partial \tau} + \Pi \bar{V}_e \frac{\partial \bar{p}_r}{\partial \tau} \quad (7)$$

where,

$$\begin{aligned} \bar{q}_s &= \{ \bar{Q}_s^{(1)} \quad \bar{Q}_s^{(2)} \quad \bar{Q}_s^{(3)} \quad \bar{Q}_s^{(4)} \} \\ \bar{p}_r &= \{ \bar{P}_r^{(1)} \quad \bar{P}_r^{(2)} \quad \bar{P}_r^{(3)} \quad \bar{P}_r^{(4)} \} \\ \bar{h}_r &= \{ \bar{h}_r^{(1)} \quad \bar{h}_r^{(2)} \quad \bar{h}_r^{(3)} \quad \bar{h}_r^{(4)} \} \\ \Delta \bar{h} &= \{ (\bar{h}_1^{(1)} - \bar{h}_1^{(4)}) \quad (\bar{h}_1^{(2)} - \bar{h}_1^{(1)}) \quad (\bar{h}_1^{(3)} - \bar{h}_1^{(2)}) \quad (\bar{h}_1^{(4)} - \bar{h}_1^{(3)}) \} \\ \Phi &= 6\mu UL_e / c^2 P_{s0} \\ \Theta &= 12\mu\omega DL_e \sin(\pi/4) / c^2 P_{s0} \\ \Pi &= 12\mu\omega\kappa / c^3 \end{aligned}$$

The fluid flow  $\bar{q}_s$ , injected into the bearing recesses, is resultant of the combination of the conventional lubrication supply and the active lubrication systems, whose main components are the servovalves and the type adopted is a two-stage, four-way, critical center or zero-lapped spool electrohydraulic servovalve. In a typical critical center servovalve, the output spool position is proportional to electrical signal applied to torque motor coils. Movement of the spool opens an orifice from the constant supply pressure  $P_s$  to one servovalve port and an identical orifice connects the other servovalve port to the return line to reservoir at pressure  $P_t$ . In case of having the bearing recess connected to the return line via servovalve port, the value of return pressure  $P_t$  has to be adjusted in order to assure a minimum operational pressure in the bearing recess.

The dynamics of the fluid flow through an unloaded servovalve can be described by a second order differential equation (Merrit, 1967). The coefficients of such equation, eigenfrequency  $\omega_v$ , damping factor  $\zeta_v$  and gain  $K_v$ , are obtained from servovalve manufacturers (Neal, 1974 and Edelmann, 1986). Each servovalve, working in orthogonal directions  $X$  and  $Y$ , is described mathematically as a function of the servovalve dynamic coefficients, the unloaded fluid flows  $Q_{vX}$  and  $Q_{vY}$  and the input electrical voltage signals  $U_x$  and  $U_y$  by

$$\begin{cases} \ddot{Q}_{vX} + 2\zeta_v\omega_v\dot{Q}_{vX} + \omega_v^2 Q_{vX} = \omega_v^2 K_v U_x \\ \ddot{Q}_{vY} + 2\zeta_v\omega_v\dot{Q}_{vY} + \omega_v^2 Q_{vY} = \omega_v^2 K_v U_y \end{cases} \quad (8)$$

The input signals  $U_x$  and  $U_y$  are generated as a linear combination of journal displacement and velocity sensors signals, and can be expressed as

$$\begin{cases} U_x = g_{1X} \Delta e_x + g_{2X} \Delta \dot{e}_x + g_{3X} \Delta e_y + g_{4X} \Delta \dot{e}_y \\ U_y = g_{1Y} \Delta e_y + g_{2Y} \Delta \dot{e}_y - g_{3Y} \Delta e_x + g_{4Y} \Delta \dot{e}_x \end{cases} \quad (9)$$

where

$$\begin{aligned} \Delta e_x &= e_x^* e^{i\omega t}, \quad \Delta e_y = e_y^* e^{i\omega t} \\ \Delta \dot{e}_x &= i\omega \Delta e_x, \quad \Delta \dot{e}_y = i\omega \Delta e_y \\ e_x^* &= e_1^* \cos \varphi_0 - e_0 \varphi_1^* \sin \varphi_0, \quad e_y^* = e_1^* \sin \varphi_0 + e_0 \varphi_1^* \cos \varphi_0 \end{aligned}$$

Such a linear combination of journal displacement and velocity signals leads to a PD (proportional-derivative) controller.

As shown in Fig.1a, the servovalve connected to recesses 1 and 3 is responsible for controlling the journal movement in  $X$  direction, and the other servovalve connected to recesses 2 and 4 is responsible for controlling the journal movement in  $Y$  direction. For simplicity, the analysis is presented only in  $Y$  direction.

The unloaded servovalve flow  $Q_{vY}$  can be expressed by  $Q_{vY} = Q_{vY}^* e^{i(\omega t + \varphi_Y)}$ , once it is proportional to the harmonic journal

eccentricity. It was assumed that the cross-coupling gains coefficients  $g_{3Y}$  and  $g_{4Y}$  are proportional to the direct gain coefficients  $g_{1Y}$  and  $g_{2Y}$ , respectively, and are related as follows:  $g_{3Y} = k_g g_{1Y}$  and  $g_{4Y} = k_g g_{2Y}$ , where  $k_g$  is the proportionality constant. The factor  $k_g$  only influences the cross stiffness and damping coefficients of the bearing. The inversion of one of the retrofitted displacement signals in Eq.(9) allows the simultaneously reduction of the cross-coupling stiffness coefficients. The crossed retrofitting of linear velocity signals should be avoided, once it contributes to the increase of the whirl frequency ratio (Santos and Watanabe, 2004). One can deduce the  $Q_{vY}$  expression from Eq.(8), in dimensionless form

$$\bar{Q}_{vY} = (\epsilon_Y^* + k_g \epsilon_X^*) G_Y e^{i(\omega t + \varphi_Y)} \quad (10)$$

where,

$$\begin{aligned} G_Y &= \frac{12\mu\omega_v^2 K_v}{c^2 P_{s0}} \sqrt{\frac{g_{1Y}^2 + \omega^2 g_{2Y}^2}{(\omega_v^2 - \omega^2)^2 + (2\zeta_v\omega_v\omega)^2}} \\ \varphi_Y &= \text{tg}^{-1} \left\{ \frac{-2\zeta_v\omega_v\omega g_{1Y} + \omega(\omega_v^2 - \omega^2)g_{2Y}}{(\omega_v^2 - \omega^2)g_{1Y} + 2\zeta_v\omega_v\omega^2 g_{2Y}} \right\} \end{aligned}$$

The active fluid flow injected in  $Y$  direction  $Q_Y$  may be given in the following dimensionless form

$$\bar{Q}_Y = \bar{Q}_{vY} - \bar{K}_{PQ} \bar{P}_{LY} \quad (11)$$

where,

$$\begin{aligned} K_{PQ} &= (\partial Q_Y / \partial P_{LY})_{op}, \quad \bar{K}_{PQ} = 12\mu K_{PQ} / c^3 \\ \bar{P}_{LY} &= P_{LY} / P_{s0} = \bar{P}_s^{(2)} - \bar{P}_s^{(4)} \equiv \Delta \bar{P}_s^{(2)} - \Delta \bar{P}_s^{(4)} \end{aligned}$$

The coefficients  $K_v$  and  $K_{PQ}$  may be experimentally determined as shown by Merrit (1967), and usually are given by servovalve manufacturers. Equation (11) is valid only in a small range of the nominal input signal, usually  $\pm 5\%$ . It is important to mention that the gains  $g_1$  and  $g_2$  have to be chosen respecting the linear range of the servovalve dynamics, i.e. the maximum amplitude of the control voltage (servovalve input signal). Mathematically, it means that  $U_x$  and  $U_y$ , given by Eq.(9), have to be in the range of  $\pm 5\%$  of the nominal voltage. Such an analysis can only be done knowing the vibration amplitudes of the rotor-bearing system.

Generally speaking, as it can be seen in Eq.(9), the bigger the rotor displacements amplitudes  $e_x^*$  and  $e_y^*$  are, the smaller the proportional coefficients  $g_1$  and  $g_2$  can be, in order to avoid a large amplitude of control signals  $U_x$  and  $U_y$ , and the resulting saturation of servovalve dynamics. Furthermore, the higher the perturbation frequencies  $\omega$  are, the smaller the derivative gain coefficient  $g_2$  can be, if a controller with derivative part is adopted, once the journal velocities amplitudes are given by  $\omega e_x^*$  and  $\omega e_y^*$ .

Coupling the HJB and servo control system models, results in the complete mathematical model of AHJB (Santos and Watanabe, 2003), and the total fluid flow injected into the bearing recesses are determined as follows

$$\bar{q}_s = \bar{q}_{s0} + (\delta_c \bar{p}_{r1} + \bar{g}_{s1}) \epsilon_1^* e^{it} + (\delta_c \bar{p}_{r2} + \bar{g}_{s2}) \epsilon_0 \varphi_1^* e^{it} \quad (12)$$

where,  $\bar{q}_{s0}$ ,  $\bar{p}_{r1}$  and  $\bar{p}_{r2}$  are defined similarly to  $\bar{q}_s$  and  $\bar{p}_r$ , and  $\bar{g}_{s1}$  and  $\bar{g}_{s2}$  are injection vectors related to the bearing active lubrication system.

Defining the following small amplitude perturbations form for  $\bar{h}_r$ ,  $\Delta\bar{h}$  and  $\bar{Q}_r$  as follows

$$\begin{cases} \bar{h}_r = \bar{h}_{r0} + \varepsilon_i^* e^{i\tau} \cos \theta_r + \varepsilon_o \phi_i^* e^{i\tau} \sin \theta_r \\ \Delta\bar{h} = \Delta\bar{h}_0 + \varepsilon_i e^{i\tau} \Delta\bar{h}_1 + \varepsilon_o \phi_i e^{i\tau} \Delta\bar{h}_2 \\ \bar{Q}_r = \bar{Q}_{r0} + \varepsilon_i e^{i\tau} \bar{Q}_{r1} + \varepsilon_o \phi_i e^{i\tau} \bar{Q}_{r2} \end{cases} \quad (13)$$

where,

$$\bar{Q}_{r1} = \left. \frac{\partial \bar{Q}_r}{\partial \varepsilon} \right|_{\varepsilon=\varepsilon_0} ; \bar{Q}_{r2} = \left. \frac{1}{\varepsilon_0} \frac{\partial \bar{Q}_r}{\partial \phi} \right|_{\phi=\phi_0}$$

Substituting Eq.(12-13) in Eq.(7), and collecting the similar linear terms, results the three following equations

$$\bar{q}_{s0} = \delta_c (I_{d,s1} - \bar{p}_{r0}) = \bar{Q}_{r0} \bar{p}_{r0} + \Phi \Delta\bar{h}_0 \quad (14)$$

$$(\bar{Q}_{r0} + \delta_c I_{d,s1} + i\Pi \bar{V}_e) \bar{p}_{r1} = -\bar{Q}_{r1} \bar{p}_{r0} - \Phi \Delta\bar{h}_1 - i\Theta \cos \theta_r + \bar{g}_{s1} \quad (15)$$

$$(\bar{Q}_{r0} + \delta_c I_{d,s1} + i\Pi \bar{V}_e) \bar{p}_{r2} = -\bar{Q}_{r2} \bar{p}_{r0} - \Phi \Delta\bar{h}_2 - i\Theta \cos \theta_r + \bar{g}_{s2} \quad (16)$$

The steady state recess pressure vector  $\bar{p}_{r0}$  is obtained solving Eq.(14), and the dynamic recess pressures  $\bar{p}_{r1}$  and  $\bar{p}_{r2}$  vectors are obtained by solving Eq.(15) and Eq.(16), respectively.

### AHJB Dynamic Coefficients

The stiffness and damping coefficients of the AHJB are determined by considering the dynamic restoring forces due to small amplitude perturbed film pressure  $\varepsilon_i^* e^{i\omega t} P_1$  and  $\varepsilon_o \phi_i^* e^{i\omega t} P_2$  around the steady state journal equilibrium position, as presented by Ghosh et al. (1989). Considering the perturbed film pressure  $\varepsilon_i^* e^{i\omega t} P_1$ , the restoring forces along  $\varepsilon_o$  and  $\phi_o$  directions, denoted by  $W_{1\varepsilon}$  and  $W_{1\phi}$ , respectively, also can be written as a function of linearized stiffness and damping coefficients. Defining the dimensionless force  $\bar{W}_1 = W_1 / LDP_{s0} \varepsilon_i^*$  and the perturbation frequency ratio  $\bar{\omega} = \omega / \Omega$ , the following dimensionless stiffness ( $\bar{K}_{\varepsilon\varepsilon}$ ,  $\bar{K}_{\phi\varepsilon}$ ) and damping ( $\bar{B}_{\varepsilon\varepsilon}$ ,  $\bar{B}_{\phi\varepsilon}$ ) coefficients can be deduced

$$\begin{cases} \bar{K}_{\varepsilon\varepsilon} = -Re\{\bar{W}_{1\varepsilon}\} = \frac{K_{\varepsilon\varepsilon}c}{LDP_{s0}} & \bar{K}_{\phi\varepsilon} = -Re\{\bar{W}_{1\phi}\} = \frac{K_{\phi\varepsilon}c}{LDP_{s0}} \\ \bar{B}_{\varepsilon\varepsilon} = -\frac{Im\{\bar{W}_{1\varepsilon}\}}{\bar{\omega}} = \frac{B_{\varepsilon\varepsilon}c\Omega}{LDP_{s0}} & \bar{B}_{\phi\varepsilon} = -\frac{Im\{\bar{W}_{1\phi}\}}{\bar{\omega}} = \frac{B_{\phi\varepsilon}c\Omega}{LDP_{s0}} \end{cases} \quad (17)$$

Similarly, considering the perturbed film pressure  $\varepsilon_o \phi_i^* e^{i\omega t} P_2$ , the restoring forces along  $\varepsilon_o$  and  $\phi_o$  directions, denoted by  $W_{2\varepsilon}$  and  $W_{2\phi}$ , respectively, the others resultant stiffness ( $\bar{K}_{\phi\phi}$ ,  $\bar{K}_{\varepsilon\phi}$ ) and damping ( $\bar{B}_{\phi\phi}$ ,  $\bar{B}_{\varepsilon\phi}$ ) coefficients, by using the dimensionless force  $\bar{W}_2 = W_2 / LDP_{s0} \varepsilon_o \phi_i^*$ , are given as follows

$$\begin{cases} \bar{K}_{\varepsilon\phi} = -Re\{\bar{W}_{2\varepsilon}\} = \frac{K_{\varepsilon\phi}c}{LDP_{s0}} & \bar{K}_{\phi\phi} = -Re\{\bar{W}_{2\phi}\} = \frac{K_{\phi\phi}c}{LDP_{s0}} \\ \bar{B}_{\varepsilon\phi} = -\frac{Im\{\bar{W}_{2\varepsilon}\}}{\bar{\omega}} = \frac{B_{\varepsilon\phi}c\Omega}{LDP_{s0}} & \bar{B}_{\phi\phi} = -\frac{Im\{\bar{W}_{2\phi}\}}{\bar{\omega}} = \frac{B_{\phi\phi}c\Omega}{LDP_{s0}} \end{cases} \quad (18)$$

The bearing stiffness and damping coefficients may be used in rotor dynamic calculations for unbalance and random vibration

responses, for determining damped critical speeds and for rotor stability analysis. For these purposes, representing the bearing dynamic coefficients in the XY inertial coordinate system is more suitable than in  $\varepsilon\phi$  reference system. Equation (19) provides the dynamic coefficients conversion between these two reference coordinate systems.

$$\begin{bmatrix} \bar{K}_{XX} & \bar{K}_{XY} \\ \bar{K}_{YX} & \bar{K}_{YY} \end{bmatrix} = T' \begin{bmatrix} \bar{K}_{\varepsilon\varepsilon} & \bar{K}_{\varepsilon\phi} \\ \bar{K}_{\phi\varepsilon} & \bar{K}_{\phi\phi} \end{bmatrix} T \quad (19)$$

$$\begin{bmatrix} \bar{B}_{XX} & \bar{B}_{XY} \\ \bar{B}_{YX} & \bar{B}_{YY} \end{bmatrix} = T' \begin{bmatrix} \bar{B}_{\varepsilon\varepsilon} & \bar{B}_{\varepsilon\phi} \\ \bar{B}_{\phi\varepsilon} & \bar{B}_{\phi\phi} \end{bmatrix} T$$

$$\text{where, } T = \begin{bmatrix} \cos \phi_0 & \sin \phi_0 \\ -\sin \phi_0 & \cos \phi_0 \end{bmatrix}$$

### Numerical Analysis and Results

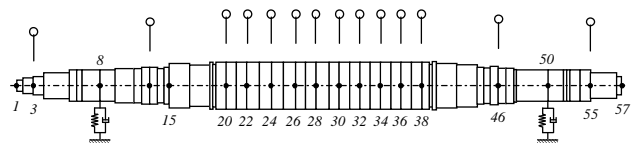
The bearing and servovalve characteristics adopted in the numerical simulations of the compressor dynamic response are presented in Tab.1.

**Table 1. AHJB geometric and operational characteristics adopted for the rotor-bearing simulations.**

journal diameter	$D = 100$	mm
bearing width	$L = 45$	mm
circumferential land width	$a = 9$	mm
axial land width	$b = 10$	mm
recess depth	$t_r = 1$	mm
nominal bearing gap	$c = 90$	$\mu\text{m}$
constant recess supply pressure range	$P_{s0} = 1.5 \text{ to } 4.0$	MPa
servovalve supply pressure	$P_s = 2 P_{s0}$	MPa
lubricant oil dynamic viscosity (37.8 °C)	$\mu = 0,015$	Ns/m <sup>2</sup>
lubricant oil density	$\rho = 900$	kg/m <sup>3</sup>
centered recess pressure ratio	$\beta = 0,5$	---
capillary dimensionless parameter	$\delta_c = 17.453$	---
servovalve eigenfrequency	$\omega_v = 2\pi.320$	rad/s
servovalve damping factor	$\zeta_v = 0.55$	---
servovalve gain	$K_v = 1 \times 10^{-5}$	m <sup>3</sup> /s/V
servovalve linear factor	$K_{PQ} = 1.8 \times 10^{-12}$	Pa/m <sup>3</sup> /s

By applying the shaft and disk finite elements modeling technique proposed by Nelson and McVaugh (1976), the gas compressor shaft is modeled with 56 shaft elements and 57 nodes, as illustrated in Fig.2.

Impellers and other machine elements attached to the shaft are considered as rigid discs, whose dynamics are incorporated into the model by adding inertia to respective nodes. Hence, in the model, the impeller are at nodes 20, 24, 28, 32 and 36; bushes are at nodes 22, 26, 30 and 34; a thrust disc sleeve is located at node 3; a balance piston is located at node 38; seal bushes are located are nodes 12 and 46, and the coupling is at node 55. The shaft bearings are located at nodes 8 and 50.



**Figure 2. Rotor compressor finite elements model.**

The HJB static and dynamic characteristics were extensively investigated by considering different design and operational parameters, characterized by the Sommerfeld number (Watanabe, 2003). Figure 3 shows the journal bearing static dimensionless pressure  $\bar{P}_0$  distribution on the land surfaces and recesses due to the static load resulting from half of the compressor weight. The static dimensionless pressures  $\bar{P}_{r0}^{(n)}$  in the recesses 1,2,3 and 4 are: 0.3389, 0.4680, 0.7194 and 0.4715, respectively.

The feasibility of influencing the AHJB dynamic fluid film coefficients by the feedback control system gains, aiming at improving the stability of a very simple rotor-bearing system was investigated in Santos and Watanabe (2003). The compensation of the cross-coupling stiffness coefficients and the increase of the direct damping coefficients were proposed in Santos and Watanabe (2004), what leads to rotor-bearing systems with larger threshold of stability. Such results are also used in the present paper.

As mentioned before, the servovalve input signals  $U_x$  and  $U_y$ , resulting from the gain and feedback signals combination, have to be in the range of  $\pm 5\%$  of the nominal voltage. Besides, bearing dynamic coefficients are only theoretically valid for infinitesimal displacements. However, according to Lund and Thomsen (1978), dynamic coefficients may be used in practical applications for amplitudes up to 50% of the bearing clearance.

Based on these information, the following restrictions of control signal and vibration amplitude were applied to the dynamic coefficients calculations: assuming that the servovalve nominal input signal range is  $U_{x,y} = \pm 10V$ , the linear actuating signals amplitudes  $U_{x,y}^*$  were restricted to  $\pm 0.25V$  range, and the dynamic displacements amplitudes  $e^*$  in 20% of nominal bearing gap, i.e.,  $e^* \leq 0.2c$ , resulting

$$\begin{cases} \frac{U_x^*}{e^*} = (1 + k_g) \sqrt{g_{1X}^2 + \omega^2 g_{2X}^2} \leq \frac{0,25}{0,2c} \\ \frac{U_y^*}{e^*} = (1 + k_g) \sqrt{g_{1Y}^2 + \omega^2 g_{2Y}^2} \leq \frac{0,25}{0,2c} \end{cases} \quad (20)$$

The influence of the feedback control gains in the bearing dynamic coefficients is investigated considering synchronous perturbation frequency due to rotor unbalance. Moreover, it is assumed that  $g_1 = g_{1X} = g_{1Y}$  and  $g_2 = g_{2X} = g_{2Y}$ , and  $k_g$ ,  $g_1$  and  $g_2$  gains are individually varied. It was observed that due to servovalve working limitations, the retrofit of velocity signals in high frequencies rapidly leads to the saturation of servovalve dynamics. Thus, only the proportional gain  $g_1 = g_{1X} = g_{1Y}$  will be varied. Assuming that  $g_2 = 0$  and  $k_g = 2$ , the dynamic coefficients of the AHJB were determined as a function of the gain  $g_1$  and the rotation frequency  $f_r$  for different values of supply pressure  $P_{s0}$ .

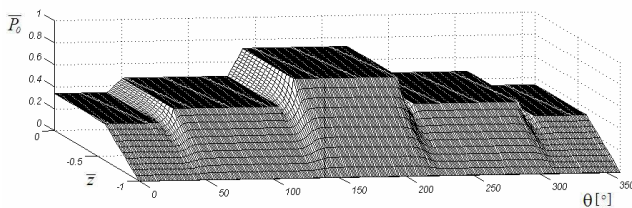


Figure 3. HJB dimensionless static pressure  $\bar{P}_0$  distribution, for  $\epsilon_0 = 0.25$  and  $\Lambda = 2$ .

Fig.4 to Fig.7 illustrate the stiffness and damping coefficients behavior of the AHJB for  $P_{s0} = 1.5$  and  $3.0MPa$ . One can clearly notice the influence of rotation frequency and control gain  $g_1$  on the bearing stiffness and damping coefficients. For a given rotation frequency  $f_r$ , and varying the control gain  $g_1$ , may be noticed three different regions in all figures: two of these regions at the extremities of gain axis  $g_1$  present constant values of the bearing dynamic coefficients and indicate the saturation of the servovalve dynamics. The third region is located in the middle of the gain axis and illustrates the variation of the dynamic coefficients as a function of the gain  $g_1$ . For a given frequency  $f_r$  such variation can nearly be described by a linear function.

Comparing Figs. 4 and 5 with Figs. 6 and 7, respectively, it can be noticed that the inclination of such linear function is strongly dependent on the supply pressure  $P_{s0}$ . Simulation results show that the higher the supply pressure  $P_{s0}$  is, the more inclined such a linear behavior is, indicating the feasibility of obtaining large changes of the dynamic coefficients with the increase of  $P_{s0}$ . With the increase of rotation frequency  $f_r$  such inclination reduces and the curve becomes more flat, indicating the impossibility of achieving significant changes of the dynamics coefficients in high frequencies due to the limitation of the servovalve dynamic response. Summarizing, if large changes of the bearing coefficients are desired and a saturation of the servovalve dynamics is achieved due to the amplitude of the control signal, an optional solution is to increase the supply pressure  $P_{s0}$ . Moreover, if the bearing coefficients shall be changed in high frequencies, higher response servovalve shall also be used.

Bearing dynamic coefficients, like those presented in Fig.4 to Fig.7, may be used in the gas compressor finite element model to analyze the rotor dynamic behavior. The rotor unbalance response and the damping ratio  $\zeta$  associated to the first two bending modes are analyzed in the case of bearing hybrid lubrication regime (HJB -  $g_1 = g_2 = 0$ ) and the active lubrication regime (AHJB), assuming  $g_2 = 0$ ,  $g_1 = 9 \times 10^3 V/m$  and  $k_g = 2$ . The criteria for choosing a suitable value for control gain  $g_1$  is based on the increasing of damping factor of the first two bending modes of the rotor in X and Y directions. It is worth to mention that such values were carefully chosen without destabilizing the higher bending mode shapes.

Fig.8 illustrates the behavior of the damping ratio  $\zeta$  of the first two rotor bending modes, as a function of the rotation frequency, for two different values of the bearing supply pressure  $P_{s0}$  (1.5 and 4.0MPa) when the bearing operates in the passive (HJB) or active (AHJB) lubrication modes. By analyzing these two graphics, one can notice a significant reduction of the damping ratio  $\zeta$  as a function of the increase of the rotation frequency  $f_r$  and of the supply pressure  $P_{s0}$ .

The increase of the rotation frequency leads to instability problems when  $\zeta < 0$ . In the case of instability the gas compressor will vibrate according to its first bending mode shape. For  $P_{s0} = 1.5MPa$ , the instability will occur at the rotation frequency of 131Hz in the case of HJB and of 151Hz in the case of AHJB. This shows that the active lubrication, aided by a simple proportional controller with cross-coupling retrofit, can enlarge the stable operational range of the gas compressor. The same behavior can be noticed in case of higher supply pressures. In case of  $P_{s0} = 4.0MPa$ , the instability will occur at the rotation frequency of 158Hz in the case of HJB and of 177Hz in the case of AHJB.

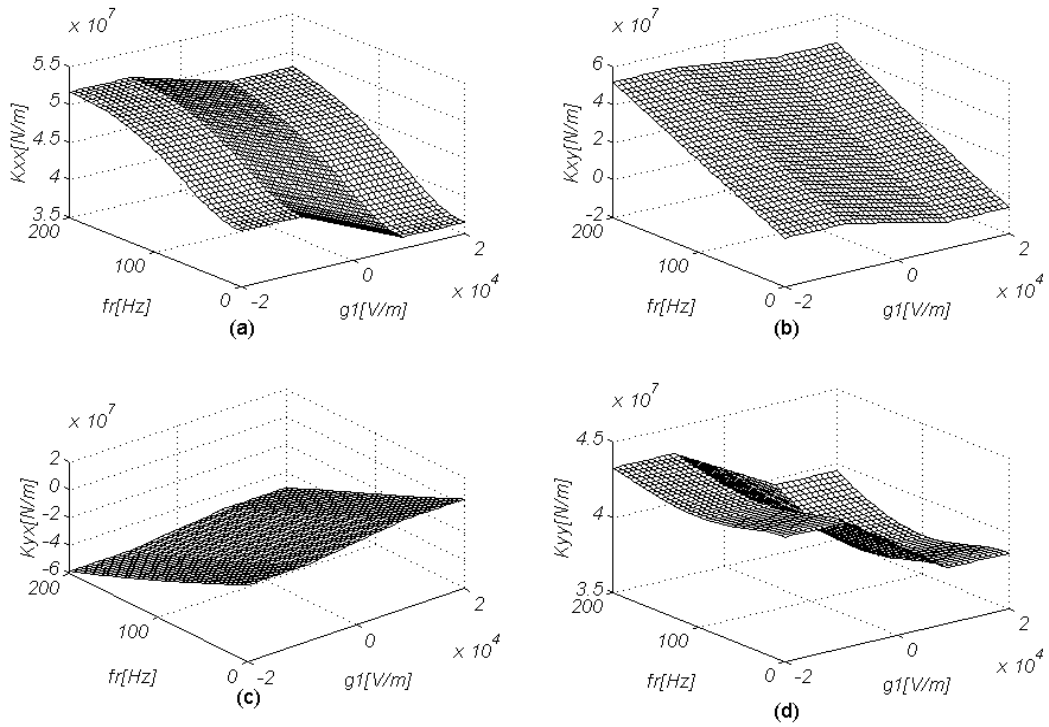


Figure 4. Bearing stiffness coefficients as function of rotation frequency  $f_r$  and proportional gain  $g_1$ , for  $P_{s0} = 1.5$ MPa.

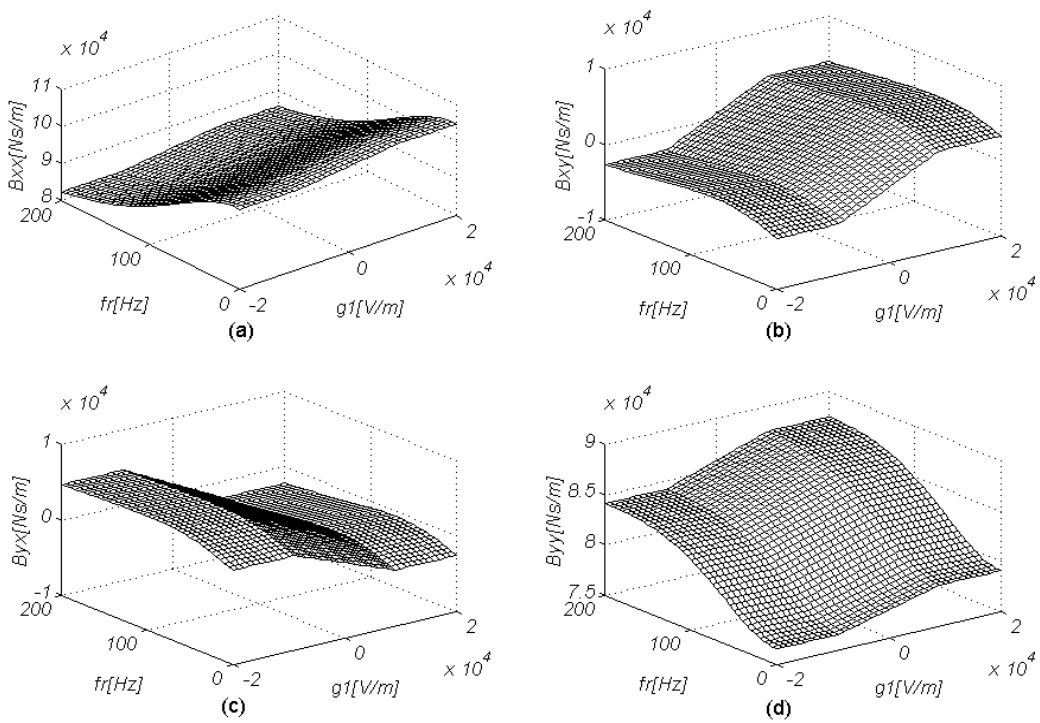


Figure 5. Bearing damping coefficients as function of rotation frequency  $f_r$  and proportional gain  $g_1$ , for  $P_{s0} = 1.5$ MPa.

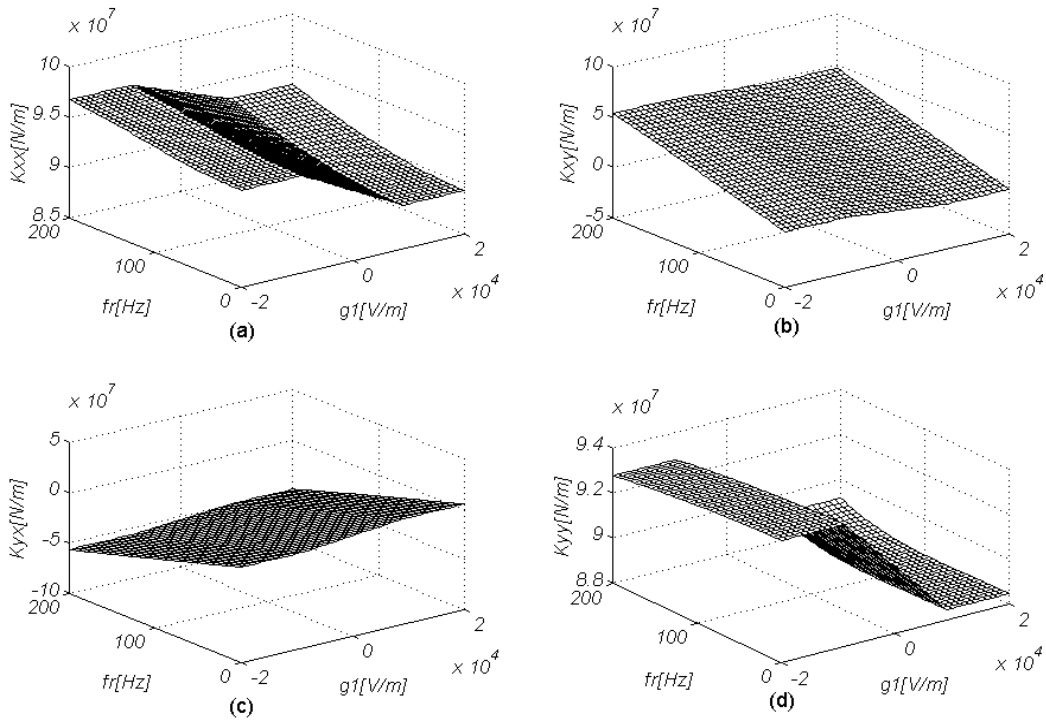


Figure 6. Bearing stiffness coefficients as function of rotation frequency  $f_r$  and proportional gain  $g_1$ , for  $P_{s0} = 3.0\text{MPa}$ .

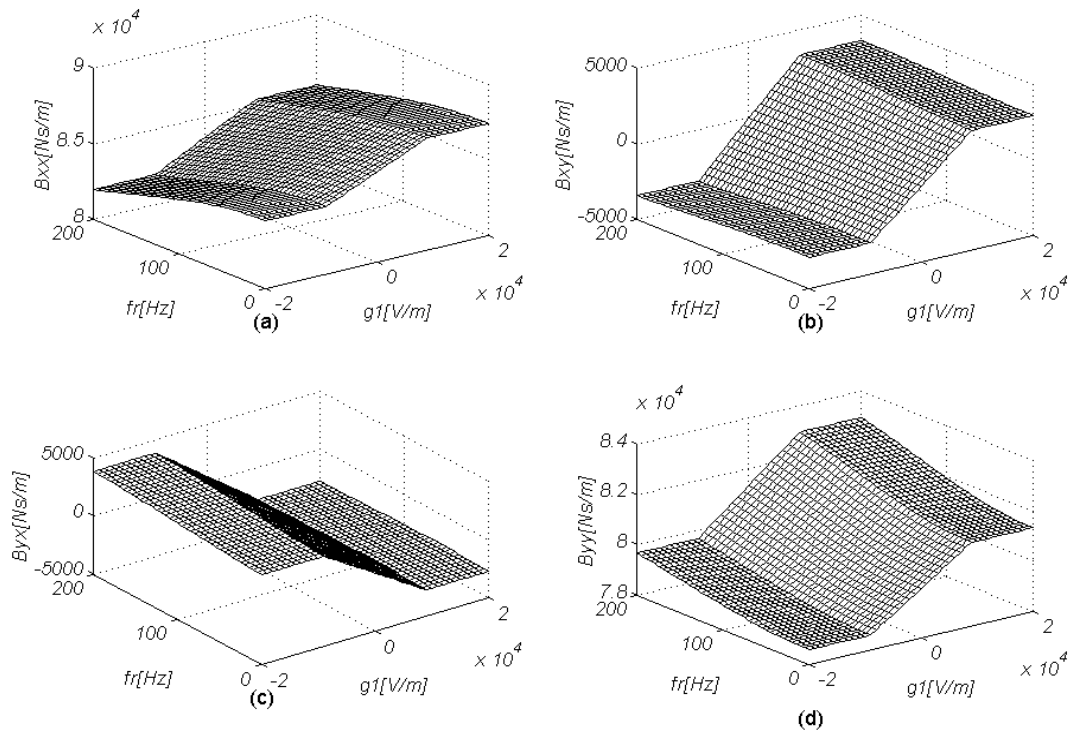


Figure 7. Bearing damping coefficients as function of rotation frequency  $f_r$  and proportional gain  $g_1$ , for  $P_{s0} = 3.0\text{MPa}$ .

Nevertheless, it is important to point out that, in the special case of multirecess bearings, an increase of the supply pressure will necessarily lead to a simultaneous increase of the stiffness coefficients of the bearing. By increasing the stiffness of the

bearings at the nodes 8 and 50, only small relative movements between compressor shaft and bearing housing will be possible in such nodes. This will deteriorate the capability of dissipation of vibration energy via squeezing the oil film in the bearings. The



consequence is a reduction of the damping ratio  $\zeta$  in the case of high supply pressures. Moreover, by increasing the supply pressure from 1.5MPa to 4.0MPa the difference between the main coefficients in the horizontal and vertical directions become less significant. Such effect can be easily seen comparing the values of  $\zeta$  in both graphics in the range of low frequencies  $f_r$ . Finally, one can claim that the values of damping ratio are reduced at low rotation frequencies by increasing the supply pressure, but the threshold of stability is increased, although the stability reserve is very low (values of  $\zeta$  very close to zero).

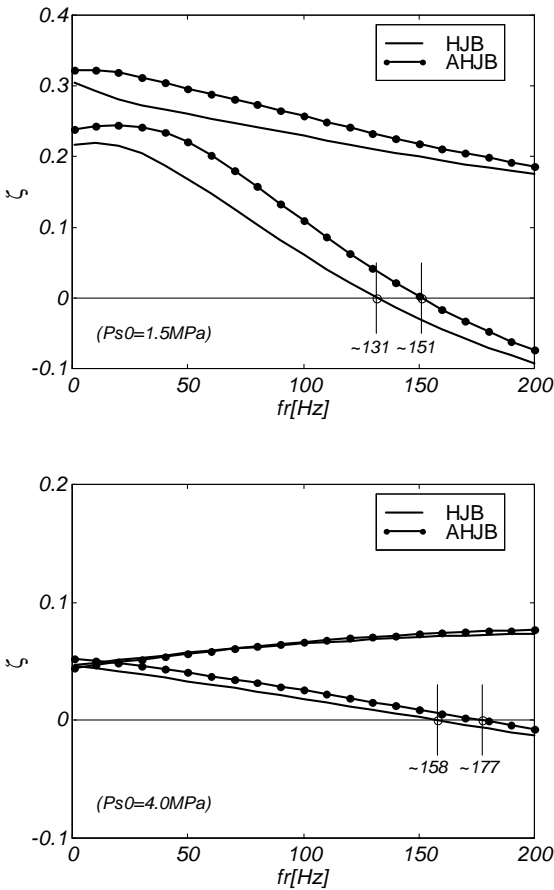


Figure 8. Damping ratio  $\zeta$  associated to the first two bending mode shapes in the case of gas compressor supported by HJB and AHJB, operating with two different supply pressures:  $P_{s0} = 1.5$  and 4.0MPa.

Applying the criteria established by the norm API 617 (1995), one can analyze the rotor unbalance response. According to this norm, for the case in study, the unbalance value to be applied in the node of maximum displacement of the model is given by  $m_u = 720g.mm$ . According to this same norm, the vibration amplitude limit for the case in study is  $L_v = 16.65\mu m$ . The unbalance response of node 28 (largest amplitude) in X and Y directions is illustrated in Fig.9 in case of hybrid operation with different values of supply pressure  $P_{s0}$ . One can notice that, for low values of  $P_{s0}$  (1.5 or 2.0MPa), the unbalance responses of the compressor are in accordance with API limit, even at critical speed, but for higher values of  $P_{s0}$  (3.0 or 4.0MPa) the unbalance response of the compressor at critical speeds are higher than the API limit. As mentioned before, in the special case of multirecess journal

bearings, an increase of the supply pressure will necessarily lead to a simultaneous increase of the stiffness coefficients of the bearing. By increasing the stiffness of the bearings at the nodes 8 and 50, only small relative movements between compressor shaft and bearing housing will be possible in such nodes. This will deteriorate the capability of dissipation of vibration energy by squeezing the oil film in the bearings and will reduce the damping ratio. Such a claim can be clearly seen in Fig.9.

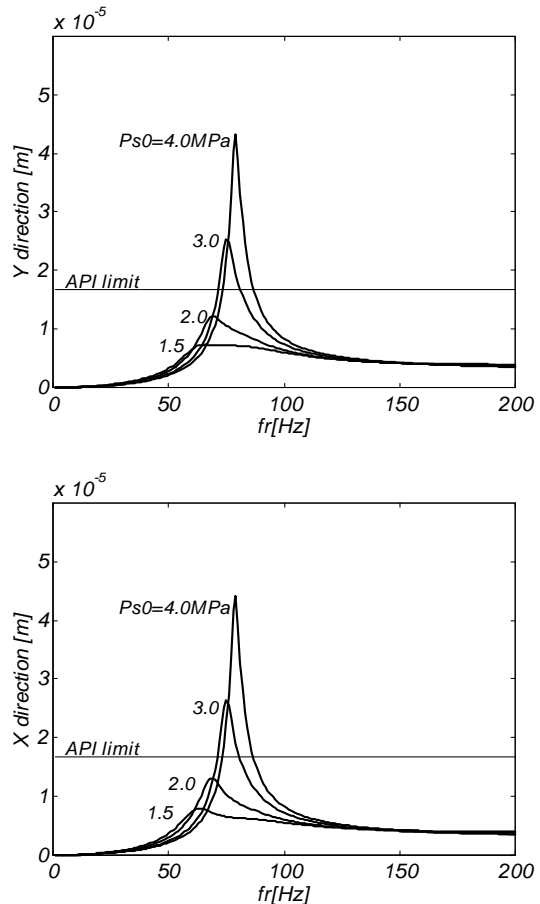


Figure 9. Unbalance response of node 28, in Y and X directions, of the compressor supported by HJB for  $P_{s0} = 1.5; 2.0; 3.0$  or 4.0MPa.

From the viewpoint of active lubrication applied to multirecess journal bearings with the aim of controlling lateral dynamics of flexible rotors a paradigm is reached: a) for improving the performance of the active lubrication in such special kind of bearings high values of supply pressure  $P_{s0}$  are needed, once the saturation of control signals retrofitted to the servovalves limits the range in which the bearing dynamics coefficients can be modified. Nevertheless, by increasing the supply pressure the main stiffness coefficients of the bearing will also increase; b) for reducing lateral vibrations of the flexible gas compressor it is very important to allow relative large amplitudes of vibration in the bearings, in order to increase the dissipation of vibration energy by squeezing the oil film. It means that it is important to keep low values of main stiffness coefficients, what coerces into working with low values of supply pressure. Fig. 8 presents the comparison of the passive (HJB) and active (AHJB) lubrication modes in the rotor unbalance response for an intermediate value of the supply pressure  $P_{s0}$

(2.5MPa), in X and Y directions, and for both, the API limit is obeyed only with the active lubrication mode. Nevertheless the improvement in reducing the unbalance response is not extraordinary significant, because the supply pressure used is relatively low. Moreover, comparing Fig.9 and Fig.10 one can conclude that a significant reduction of the unbalance response of node 28 can easily be achieved by operating in the (passive) hybrid lubrication mode with low supply pressures in the order of 1.5 and 2.0MPa.

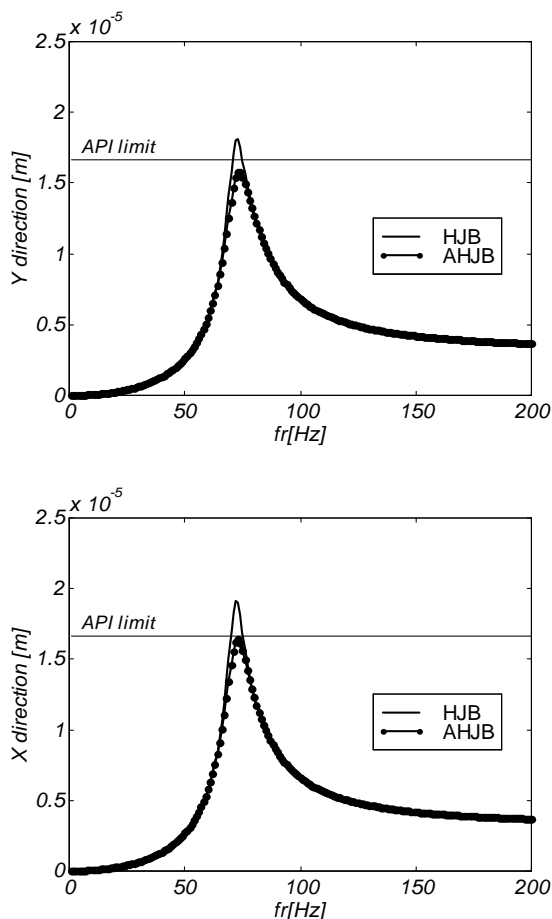


Figure 10. Unbalance response of node 28, in Y and X directions, of the compressor supported by HJB and AHJB, for  $P_{s0} = 2.5\text{MPa}$ .

## Conclusions

The mathematical modeling for calculating the static and dynamic characteristics of hybrid multirecess journal bearings, under hybrid and active lubrication regimes, was presented. The feasibility of influencing the oil film dynamic coefficients of such a kind of bearing by means of the servo control system was theoretically illustrated. The feasibility of controlling lateral dynamics of a flexible gas compressor and enlarging its stable operational range were investigated. The active lubrication, aided by a simple proportional controller with cross-coupling retrofit, can enlarge the stable operation range of the gas compressor. In case of supply pressure of  $P_{s0} = 1.5\text{MPa}$  it is possible to shift the threshold of instability in approx. 15%, from 131Hz to 151Hz. In case of  $P_{s0} = 4.0\text{MPa}$ , it is possible to shift the threshold of instability in

approx. 12%, from 158Hz to 177 Hz. Once the stiffness coefficients are strongly influenced by the bearing supply pressure  $P_{s0}$ , a suitable choice of the control gains is necessary to comply with the design requirements, unbalance response limits and threshold of instability.

Nevertheless, it is important to point out that, in the special case of multirecess journal bearings, an increase of the supply pressure will necessarily lead to simultaneous increase of the main stiffness coefficients of the bearing. By increasing the stiffness of the bearings, only small relative movements between shaft and bearing housing will be possible. This will deteriorate the capability of dissipation of vibration energy by squeezing the oil film in the bearings and will reduce the damping ratio associated to the first bending mode of the gas compressor. Moreover, a more significant and sufficient improvement of the rotor unbalance response can be achieved by only changing the supply pressure and operating the HJB in its passive form. It is not necessary to operate the bearing in the active form.

In contrast to multirecess journal bearings, tilting-pad journal bearings allow a reduction of the main stiffness of the bearing when the supply pressure is increased due to the tilt movement of the pads, Santos et al. (2004). Thus, the application of active lubrication to tilting-pad journal bearings becomes more efficient than the application to multirecess journal bearings. The reduction of the main stiffness coefficients by increasing the supply pressure allows an optimal improvement of lateral dynamics of flexible rotating shaft via active lubricated tilting-pad bearings.

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