



## A General Symplectic Method for the Response Analysis of Infinitely Periodic Structures Subjected to Random Excitations

### Abstract

A general symplectic method for the random response analysis of infinitely periodic structures subjected to stationary/non-stationary random excitations is developed using symplectic mathematics in conjunction with variable separation and the pseudo-excitation method (PEM). Starting from the equation of motion for a single loaded substructure, symplectic analysis is firstly used to eliminate the dependent degrees of the freedom through condensation. A Fourier expansion of the condensed equation of motion is then applied to separate the variables of time and wave number, thus enabling the necessary recurrence scheme to be developed. The random response is finally determined by implementing PEM. The proposed method is justified by comparison with results available in the literature and is then applied to a more complicated time-dependent coupled system.

### Keywords

Infinitely periodic structure; Symplectic mathematics; Variable separation; Pseudo-excitation method; Random vibration

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## 1 INTRODUCTION

Infinitely periodic structures are widely used in engineering practice, e.g. railway tracks, multi-span bridges and petroleum pipe-lines. They consist of identical substructures that are joined together to form a continuous structure. In recent decades, much attention has been paid to such structures and many important advances have been made, mainly in the areas of vibration characteristics, free vibration propagation and forced vibration induced by stationary harmonic loads [4, 11–16, 18, 19, 21, 22, 24, 25]. In particular, symplectic mathematics has been applied successfully [21, 22, 24, 25] to provide a precise and efficient approach for investigating the dynamic response and wave propagation caused by harmonic forces. Subsequently Lin *et al.* derived the stationary/non-stationary random response by means of the pseudo-excitation method (PEM) [5–8] and Lu *et al.* [9] applied this work to the random vibration analysis of coupled vehicle-track systems with the fixed-vehicle model, which considerably reduced the number of degrees of freedom (DOFs) required to describe the track.

15 However, vibration of infinitely periodic structures subjected to arbitrary excitation has  
16 received much less attention. Belotserkovskiy [1] investigated an infinitely periodic beam sub-  
17 jected to a moving harmonic load by analyzing one beam segment between neighboring sup-  
18 ports with boundary conditions derived from Bernoulli-Euler beam theory and this was later  
19 extended to deal with infinitely periodic strings [2, 3] ; Sheng *et al.* [17] proposed a wave  
20 number-based approach to study a two-and-a-half-dimensional finite-element model subjected  
21 to a moving or stationary harmonic load, while Mead's [10] latest advance presents a general  
22 theory for the forced vibration of multi-coupled, one-dimensional periodic structures by firstly  
23 analyzing the semi-infinite periodic system excited only at its end, which is then connected to  
24 either side of the loaded substructure. The present authors [20], based on the work of Lu *et*  
25 *al.* [9], selected a series of wave numbers evenly distributed in the interval  $[0, 2\pi)$  and derived  
26 the corresponding propagation constants. This enabled the random response of the infinitely  
27 periodic structures to be obtained by accumulating the pass-band frequency responses. Such  
28 an approach, when combined with PEM, results in an efficient method for computing response  
29 PSDs of vehicle-track coupled systems based on the moving-vehicle model. However, one draw-  
30 back stems from the discreteness of wave numbers, which inevitably causes discrete numerical  
31 errors.

32 In order to eliminate this problem and substantially improve the technique, a continuous  
33 integration is used as follows in this paper yield to a new and general approach for the response  
34 analysis of infinitely periodic structures subjected to arbitrary excitations. This new method  
35 is based on a symplectic mathematical scheme combined with a variable separation approach  
36 in which only the loaded substructure is included in the calculation. The dependent DOFs are  
37 firstly condensed into the independent ones according to the properties of the wave propagation  
38 constants. The condensed equation of motion is then derived, in which the coefficient matrices  
39 are functions of the wave number. By applying Fourier expansions to these coefficient matrices  
40 and the response vectors, the time and wave number variables are easily separated and a  
41 recurrence scheme is developed accordingly. Finally, in accordance with the work of Lin *et al.*[5,  
42 6], the resulting equations are combined with PEM for stationary or non-stationary random  
43 response analysis, after which the response power spectral densities (PSDs) and the standard  
44 deviations can be derived conveniently. The proposed method is justified by comparison with  
45 a numerical example in Reference [6] and the theory is then applied to the random analysis of  
46 a mass moving on a rail that is supported periodically ad infinitum.

## 47 2 SYMPLECTIC ANALYSIS FOR AN INFINITELY PERIODIC STRUCTURE SUBJECTED 48 TO ARBITRARY LOADS

49 In this section, the symplectic mathematical scheme is generalized to investigate the response  
50 of an infinitely periodic structure subjected to arbitrary loads. The infinitely periodic structure  
51 shown in Figure 1 consists of two kinds of substructures, denoted as sub and sub\*, which are  
52 identical except that sub\* is subjected to an arbitrary load  $f(t)$ .

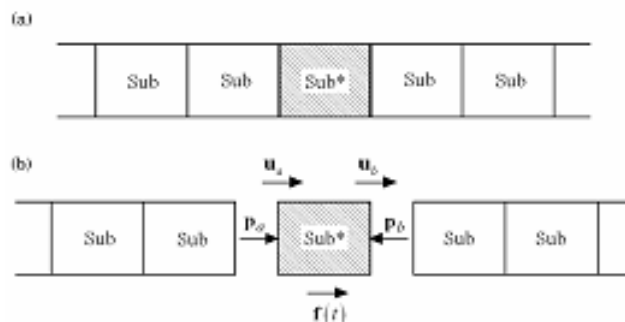


Figure 1 Infinitely periodic structure showing the loaded substructure sub\* and the forces and displacements at its interfaces with its neighbours.

53 The equation of motion for this substructure is

$$M\ddot{u} + C\dot{u} + Ku = f(t) + f_b \quad (1)$$

54 in which:  $\mathbf{M}$ ,  $\mathbf{C}$  and  $\mathbf{K}$  are the  $n \times n$  mass, damping and stiffness matrices that can be created  
55 by any means;

$$u = \{ u_a^T \quad u_b^T \quad u_i^T \}^T; f_b = \{ p_a^T \quad -p_b^T \quad 0 \}^T; \quad (2)$$

56 where: superscript T denotes transpose;  $u_a$  and  $u_b$  are the displacement vectors at the left-  
57 and right-hand interface, see Fig 1;  $u_i$  is the internal displacement vector and;  $p_a$  and  $p_b$  are  
58 the corresponding nodal force vectors on the interfaces.

59 For an undamped and unloaded substructure, it has been proven in References [21, 22, 24,  
60 25] that

$$\begin{Bmatrix} u_b \\ p_b \end{Bmatrix} = S \begin{Bmatrix} u_a \\ p_a \end{Bmatrix} = \mu \begin{Bmatrix} u_a \\ p_a \end{Bmatrix} \quad (3)$$

61 in which  $S$  is a frequency-dependent symplectic transfer matrix that has eigenvalues  $\mu$  and  
62 satisfies the symplectic orthogonality relationships

$$S^T J_n S = J_n; \quad J_n = \begin{bmatrix} 0 & I_n \\ -I_n & 0 \end{bmatrix}; \quad J_n^T = J_n^{-1} = -J_n \quad (4)$$

63 where:  $I_n$  is the  $n$ -dimensional unit matrix and; the  $\mu$  are known as the wave propagation  
64 constants, where  $|\mu| = 1$  refers to transmission waves that propagate without decay, i.e. they  
65 lie within the frequency pass-band.  $\mu$  can be expressed as

$$\mu = e^{j\theta}; \quad j = \sqrt{-1} \quad (5)$$

66 in which  $\theta$  is the wave number and lies in the interval  $[0, 2\pi)$ .

67 Let

$$T(\theta) = T = \begin{bmatrix} I_n & 0 \\ e^{j\theta} I_n & 0 \\ 0 & I_n \end{bmatrix} \quad (6)$$

68 Then for each wave number  $\theta$ , it can be verified that

$$\begin{Bmatrix} u_a^* \\ u_b^* \\ u_i^* \end{Bmatrix} = T \begin{Bmatrix} u_a^* \\ u_i^* \end{Bmatrix}; T^H \begin{Bmatrix} p_a^* \\ -p_b^* \\ 0 \end{Bmatrix} = \begin{Bmatrix} p_a^* - e^{-j\theta} p_b^* \\ 0 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \end{Bmatrix} \quad (7)$$

69 in which: superscript H denotes complex conjugate transpose;  $u_a^*$ ,  $u_b^*$  and  $u_i^*$  are the response  
 70 vectors related to a given wave number and  $p_a^*$  and  $p_b^*$  are the corresponding nodal force vectors  
 71 and hence are functions of wave number  $\theta$  and time  $t$ . Substituting Eq. (7) into Eq. (1)  
 72 and pre-multiplying both sides by  $T^H$  gives the condensed equation of motion of the loaded  
 73 substructure as

$$\bar{M}^*(\theta) \ddot{\bar{u}}^*(\theta, t) + \bar{C}^*(\theta) \dot{\bar{u}}^*(\theta, t) + \bar{K}^*(\theta) \bar{u}^*(\theta, t) = T^H(\theta) f(t) \quad (8)$$

74 in which

$$\bar{u}^* = \begin{Bmatrix} u_a^{*T} & u_i^{*T} \end{Bmatrix}^T; \bar{M}^* = T^H M T; \bar{C}^* = T^H C T; \bar{K}^* = T^H K T \quad (9)$$

75 Note that the natural frequencies of the infinitely periodic structure can be obtained by solving  
 76 the following generalized eigenproblem [16]

$$\bar{K}^* \Psi = \bar{M}^* \Psi \Omega^2 \quad (10)$$

77 in which:  $\Omega$  is the diagonal matrix of natural frequencies and  $\Psi$  is the corresponding modal  
 78 matrix. The number of natural frequencies developed from each wave number is equal to  
 79 the number of independent DOFs of the substructure. Since there are infinitely many wave  
 80 numbers, an infinitely periodic structure yields an infinite number of natural frequencies. In  
 81 Reference [20] a finite number of wave numbers, evenly distributed in the interval  $[0, 2\pi)$ ,  
 82 were selected to calculate the responses. This inevitably results in the discrete errors men-  
 83 tioned previously. However, this is circumvented below by performing a continuous integration  
 84 instead.

85 Assume that the response of each substructure can be determined by performing the fol-  
 86 lowing integration.

$$u_k(t) = \frac{1}{2\pi} \int_0^{2\pi} T(\theta) \bar{u}^*(\theta, t) e^{jk\theta} d\theta; \quad (k = 0, \pm 1, \pm 2 \dots) \quad (11)$$

87 where  $k = 0$ ,  $k > 0$ ,  $k < 0$  correspond, respectively, to the loaded substructure and the sub struc-  
 88 tures to its right and left. However,  $\bar{u}^*(\theta, t)$  cannot be solved from Eq. (8) directly and so the  
 89 following approach is used instead.

90 Let the matrices  $\bar{M}^*$ ,  $\bar{C}^*$ ,  $\bar{K}^*$  and  $T$  be expressed as

$$\begin{aligned} \bar{M}^* &= \bar{M}_0 + \bar{M}_1 e^{j\theta} + \bar{M}_{-1} e^{-j\theta}; & \bar{C}^* &= \bar{C}_0 + \bar{C}_1 e^{j\theta} + \bar{C}_{-1} e^{-j\theta} \\ \bar{K}^* &= \bar{K}_0 + \bar{K}_1 e^{j\theta} + \bar{K}_{-1} e^{-j\theta}; & T &= T_0 + T_{-1} e^{j\theta} \end{aligned} \quad (12)$$

91 in which

$$\begin{aligned} \bar{M}_0 &= \begin{bmatrix} M_{aa} + M_{bb} & M_{ai} \\ & M_{ia} \\ & & M_{ii} \end{bmatrix}; \quad \bar{M}_1 = \begin{bmatrix} M_{ab} & 0 \\ & M_{ib} \end{bmatrix}; \quad \bar{M}_{-1} = \begin{bmatrix} M_{ba} & M_{bi} \\ & 0 \end{bmatrix} \\ T_0 &= \begin{bmatrix} I_n & 0 \\ & 0 \\ & & I_n \end{bmatrix}; \quad T_{-1} = \begin{bmatrix} 0 & 0 \\ I_n & 0 \\ & 0 \end{bmatrix} \end{aligned} \quad (13)$$

92 where:  $M_{lm}$  ( $l, m = a, b, i$ ) are the submatrices corresponding, respectively, to the DOFs at the  
 93 two interfaces and the internal DOFs and; the submatrices of  $\bar{C}$  and  $\bar{K}$  are defined analogously  
 94 to those of  $\bar{M}$ . Now  $\bar{u}^*(\theta, t)$  can be expressed as the sum of an infinite number of spatial  
 95 harmonics by using Fourier expansion to give

$$\bar{u}^* = \sum_n \bar{u}_{en} e^{jn\theta}; \quad (n = 0, \pm 1, \pm 2 \dots) \quad (14)$$

96 in which  $\bar{u}_{en}$  ( $n = 0, \pm 1, \pm 2 \dots$ ) denotes the Fourier expansion coefficients. Eq. (11) can then  
 97 be rewritten as

$$u_k(t) = T_0 \bar{u}_{e(-k)} + T_{-1} \bar{u}_{e(-k-1)}; \quad (k = 0, \pm 1, \pm 2 \dots) \quad (15)$$

98 Substituting Eqs. (12) and (14) into Eq. (8) and separating the variables of time and wave  
 99 number by using the orthogonality of the exponents gives

$$\begin{bmatrix} \hat{M}_{mm} & \hat{M}_{ms} \\ \hat{M}_{sm} & \hat{M}_{ss} \end{bmatrix} \begin{Bmatrix} \ddot{\hat{u}}_{mk} \\ \ddot{\hat{u}}_s \end{Bmatrix} + \begin{bmatrix} \hat{C}_{mm} & \hat{C}_{ms} \\ \hat{C}_{sm} & \hat{C}_{ss} \end{bmatrix} \begin{Bmatrix} \dot{\hat{u}}_{mk} \\ \dot{\hat{u}}_s \end{Bmatrix} + \begin{bmatrix} \hat{K}_{mm} & \hat{K}_{ms} \\ \hat{K}_{sm} & \hat{K}_{ss} \end{bmatrix} \begin{Bmatrix} \hat{u}_{mk} \\ \hat{u}_s \end{Bmatrix} = \begin{bmatrix} F_{mk} \\ 0 \end{bmatrix} f(t) \quad (k = 1, 2, \dots) \quad (16)$$

100 in which

$$\begin{aligned} \hat{u}_{mk} &= \{ \bar{u}_{e0}^T \quad \bar{u}_{e1}^T \quad \bar{u}_{e-1}^T \quad \dots \quad \bar{u}_{ek}^T \quad \bar{u}_{e-k}^T \}^T; \quad \hat{u}_s = \{ \bar{u}_{e(k+1)}^T \quad \bar{u}_{e-(k+1)}^T \quad \dots \}^T; \\ \hat{M}_{mm} &= \begin{bmatrix} \bar{M}_0 & \bar{M}_{-1} & \bar{M}_1 & & & \\ \bar{M}_1 & \bar{M}_0 & \bar{M}_{-1} & & & \\ \bar{M}_{-1} & \bar{M}_1 & \bar{M}_0 & & & \\ & & & \ddots & & \\ & & & & \bar{M}_{-1} & \\ & & & & & \bar{M}_0 \end{bmatrix}; \quad F_{mk} = \begin{bmatrix} T_0^T \\ 0 \\ T_{-1}^T \\ 0 \\ \vdots \\ 0 \end{bmatrix} \\ \hat{M}_{ss} &= \begin{bmatrix} \bar{M}_0 & & \bar{M}_{-1} & & & \\ & \bar{M}_0 & & \bar{M}_1 & & \\ \bar{M}_1 & & \bar{M}_0 & & \ddots & \\ & \bar{M}_{-1} & & \bar{M}_0 & & \\ & & & \ddots & & \ddots \end{bmatrix}; \quad \hat{M}_{ms} = \begin{bmatrix} 0 & & \dots & 0 \\ \vdots & & & \vdots \\ 0 & & \ddots & \vdots \\ \bar{M}_{-1} & 0 & & \vdots \\ & \bar{M}_1 & 0 & \dots & 0 \end{bmatrix} = M_{sm}^T \end{aligned} \quad (17)$$

101 and  $\hat{C}$  and  $\hat{K}$  can be substituted for  $\hat{M}$  throughout Eq. (17).

102 By inspection it can be seen that:  $\hat{u}_{mk}$  is a finite-dimensional vector;  $\hat{u}_s$  is of infinite-  
 103 dimension and; the matrices of Eq. (16) are very sparse. Thus for each value of  $k$ , Eq. (16)  
 104 can be rewritten in block form as

$$\hat{M}_{mm}\ddot{\hat{u}}_{mk} + \hat{C}_{mm}\dot{\hat{u}}_{mk} + \hat{K}_{mm}\hat{u}_{mk} = F_{mk}f(t) - M_{ms}\ddot{u}_s - C_{ms}\dot{u}_s - K_{ms}u_s \quad (18)$$

105

$$M_{ss}\ddot{u}_{sk} + C_{ss}\dot{u}_{sk} + K_{ss}u_{sk} = -M_{sm}\ddot{u}_{mk} - C_{sm}\dot{u}_{mk} - K_{sm}u_{mk} \quad (19)$$

106 where:  $u_s = \{ \bar{u}_{e(k+1)}^T \quad \bar{u}_{e-(k+1)}^T \}^T$ ;  $u_{mk} = \{ \bar{u}_{ek}^T \quad \bar{u}_{e-k}^T \}^T$  and;  $M_{ms}$ ,  $M_{sm}$ ,  $C_{ms}$ ,  $C_{sm}$ ,  $K_{ms}$   
 107 and  $K_{sm}$  are submatrices. Noting that Eq. (19) is of infinite-dimension, it needs to be  
 108 calculated in truncated form. Since its coefficient matrices remain unchanged irrespective of  
 109 the value of  $k$ , Eq. (19) can be transformed into state-space as [23]

$$\dot{v}_s = H_s v_s + Q v_{mk} \quad (20)$$

110 in which  $v_s = \{ \hat{u}_s^T \quad \dot{\hat{u}}_s^T \}^T$ ;  $v_{mk} = \{ u_{mk}^T \quad \dot{u}_{mk}^T \}^T$ ;  $H_s$  is a Hamiltonian matrix and;  $Q$  is the  
 111 load coefficient matrix. Usually, Eq. (20) is solved using a step-by-step integration scheme.  
 112 Thus if the response at time  $t$  is known, the response at time  $t + \Delta t$  can be expressed as

$$v_s(t + \Delta t) = T_s(\Delta t) v_s(t) + R v_{mk}(t) \quad (21)$$

113 where  $T_s(\Delta t)$  is an exponential matrix whose precise computation is described in Reference  
 114 [23] and the physical meaning of the  $n$ -th column of matrix  $R$  is the response  $v_s(t + \Delta t)$  when  
 115 assuming that  $v_s(t) = 0$  and that the  $n$ -th value of  $v_{mk}(t)$  is 1 while all others are zero.  
 116 Consequently, the responses can be computed by the following recurrence scheme: (1) let  $k = 1$   
 117 and solve Eqs. (18) and (21) by using step-by-step integration to obtain the responses  $\bar{u}_{e0}$ ,  $\bar{u}_{e1}$   
 118 and  $\bar{u}_{e-1}$ ; (2) Similarly, let  $k = 2$  and substitute  $\bar{u}_{e0}$ ,  $\bar{u}_{e1}$  and  $\bar{u}_{e-1}$  into Eq. (18) to obtain  $\bar{u}_{e2}$   
 119 and  $\bar{u}_{e-2}$  and; (3) Compute the remaining responses similarly and hence find the responses of  
 120 the substructures by using Eq. (15).

121 Note that the method is still applicable if the coefficient matrices of Eq. (1) are time-  
 122 dependent, e.g. due to a moving mass coupling with the infinitely periodic structure.

### 123 3 RESPONSES OF INFINITELY PERIODIC STRUCTURES SUBJECTED TO RAN- 124 DOM EXCITATIONS

125 PEM is an accurate and highly efficient algorithm for structural stationary or non-stationary  
 126 random response analysis. In this section, it is combined with the above method to find the  
 127 random responses. Consider the most complicated case of a time-dependent system excited  
 128 by an evolutionary random point excitation. Then the equation of motion of the system is

$$M\ddot{u} + C\dot{u} + Ku = f(t) + f_b \quad (22)$$

$$f(t) = r(t)g(t)x(t)$$

129 in which:  $M$ ,  $C$  and  $K$  are functions of time;  $r(t)$  identifies which element is being excited;  
 130  $g(t)$  is the modulation function and;  $x(t)$  is a stationary random process with PSD  $S_{xx}(\omega)$ .

131 The corresponding response vector can be expressed by the convolution integral

$$u(t) = \int_0^t H(t, \tau) f(\tau) d\tau \quad (23)$$

132 in which  $H(t, \tau)$  is the frequency response matrix. Multiplying  $u(t)$  by its transpose and  
 133 applying the mathematical expectation operator, the variance matrix of the response vector is  
 134 given by

$$\begin{aligned} R_{uu}(t) &= E[u(t)u^T(t)] = \int_0^t \int_0^t H(t, \tau_1) E[f(\tau_1) f^T(\tau_2)] H^T(t, \tau_2) d\tau_1 d\tau_2 \\ &= \int_0^t \int_0^t H(t, \tau_1) r(\tau_1) r^T(\tau_2) H^T(t, \tau_2) g(\tau_1) g(\tau_2) E[x(\tau_1) x(\tau_2)] d\tau_1 d\tau_2 \end{aligned} \quad (24)$$

135 According to the Wiener - Khintchine theorem

$$E[x(\tau_1) x(\tau_2)] = R_{xx}(\tau) = \int_{-\infty}^{\infty} S_{xx}(\omega) e^{i\omega(\tau_2 - \tau_1)} d\omega \quad (25)$$

136 Substituting Eq. (25) into Eq. (24) and exchanging the integral order gives the evolutionary  
 137 PSD matrix of the response vector  $u(t)$  as

$$R_{uu}(t) = \int_{-\infty}^{+\infty} S_{uu}(\omega, t) d\omega \quad (26)$$

138 where

$$S_{uu}(\omega, t) = \int_0^t \int_0^t H(t, \tau_1) r(\tau_1) r^T(\tau_2) H^T(t, \tau_2) g(\tau_1) g(\tau_2) S_{xx}(\omega) e^{i\omega(\tau_2 - \tau_1)} d\tau_1 d\tau_2 \quad (27)$$

139 It can be seen that Eq. (27) is a double integral expression which is very time consuming  
 140 to compute directly. Therefore, PEM is used instead. Assume that the structure is subjected  
 141 to a pseudo-excitation

$$\tilde{f}(\omega, t) = r(t)g(t) \sqrt{S_{xx}(\omega)} e^{i\omega t} \quad (28)$$

142 Eq. (26) can then be rewritten as

$$S_{uu}(\omega, t) = \tilde{u}^*(\omega, t) \tilde{u}^T(\omega, t); \tilde{u}(\omega, t) = \int_0^t H(t, \tau) \tilde{f}(\omega, \tau) d\tau \quad (29)$$

143 where the superscript \* denotes complex conjugate. It is clear that  $\tilde{u}(\omega, t)$  is the response of  
 144 the structure when it is subjected to the pseudo-excitation and also that the first of Eqs. (29)  
 145 has a much simpler form than Eq. (27). Thus the use of PEM to transform random excitations  
 146 into harmonic pseudo-excitations leads to a very significant reduction in computational effort.

147 Substituting the pseudo excitation of Eq. (28) into Eq. (18) enables the pseudo responses  
 148 of the infinitely periodic structure to be obtained using the above recurrence scheme. Denoting  
 149 the pseudo response of the response  $u(t)$  as  $\tilde{u}(\omega, t)$  and utilizing PEM, the PSD of  $u(t)$  can  
 150 be written as

$$S(\omega, t) = \tilde{u}(\omega, t) \tilde{u}^*(\omega, t) \quad (30)$$

151 It is clear that if  $M$ ,  $C$  and  $K$  are time-independent, the system degenerate into a time-  
 152 independent one, and if  $g(t) = 1$ , the random excitation degenerates into a stationary one.  
 153 PEM is still applicable in these cases.

154 **4 NUMERICAL EXAMPLES**

155 **4.1 Example 1: Correctness verification**

156 In this section, the proposed method is justified by comparison with the method proposed in  
 157 Reference [6].

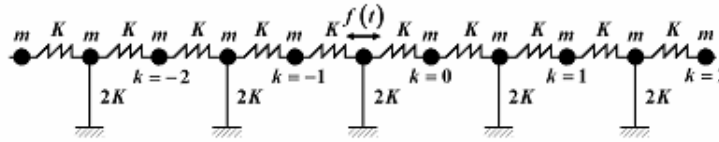


Figure 2 The infinitely periodic structure of Example 1 which is subjected to the point evolutionary random excitation  $f(t)$ . It consists of cantilever columns with stiffness  $2K$  for lateral displacements at their upper ends and which carry masses  $m$  which are connected by two springs of stiffness  $K$  with a mass  $m$  where they are connected together.

158 Consider the infinitely periodic structure defined in Figure 2 and its caption, subjected to  
 159 an evolutionary random excitation given by

$$f(t) = g(t) x(t) \tag{31}$$

160 in which the modulation function  $g(t)$  has the form shown in Figure 3, i.e.

$$g(t) = \begin{cases} 0.1t & \text{when } 0 \leq t \leq 10 \\ 1.0 & \text{when } 10 < t \leq 40 \\ 0.1(50 - t) & \text{when } 40 < t \leq 50 \\ 0 & \text{otherwise} \end{cases} \tag{32}$$

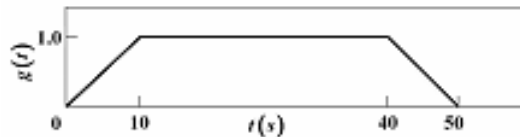


Figure 3 Envelope function  $g(t)$

161  $x(t)$  is considered as a band-limited white noise, its units being  $N^2s$

$$S_{xx}(\omega) = \begin{cases} 1.0 & \text{when } |\omega| \leq \omega_0 \\ 0.0 & \text{when } |\omega| > \omega_0 \end{cases} \tag{33}$$

162 The calculations used  $K = 1$ ;  $m = 1$ ;  $\omega_0 = 3$  and the hysteretic damping factor  $\nu = 0.1$ .

163 Figure 4 gives the time dependent variances of the displacements at stations  $k = 0, 1$  and  
 164  $2$ , with the results from the proposed method shown as the solid line, while those from the  
 165 theory of Reference [6] are given by the asterisks. Clearly the results agree very well and the  
 166 difference of the peak values at point A is less than 0.01%, which justifies the correctness of  
 167 the proposed method.



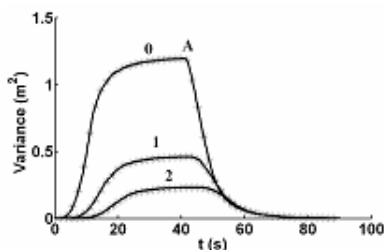


Figure 4 Time-dependent variances of the displacements at stations 0, 1 and 2 of Example 1.

168 **4.2 Example 2: Application to a time-dependent coupled system**

169 In this example, the proposed method is applied to find the time-dependent random responses  
 170 when a mass of  $1000kg$  crosses an infinite periodically supported rail/sleeper/ballast system  
 171 at a velocity of  $100km/h$ , see Figure 5. The track irregularity is regarded as white noise with  
 172 PSD  $S_{rr}(\omega) = 1.0(m^2/rad/s)$  and the parameters of the system are listed in Table 1.

Table 1 Parameters, defined in Figure 5, of the periodically supported rail of Example 2.

|                           |                          |                 |                          |
|---------------------------|--------------------------|-----------------|--------------------------|
| Bending stiffness $EI$    | $6.62 \times 10^6 N m^2$ | Stiffness $K_b$ | $1.82 \times 10^8 N/m$   |
| Rail mass/length $\rho A$ | $60.64 kg/m$             | Stiffness $K_f$ | $1.47 \times 10^8 N/m$   |
| Spacing $l$               | $0.545 m$                | Damping $C_p$   | $7.5 \times 10^4 Ns/m$   |
| Mass $M_s$                | $237 kg$                 | Damping $C_b$   | $5.88 \times 10^4 Ns/m$  |
| Mass $M_b$                | $1478 kg$                | Damping $C_f$   | $3.115 \times 10^4 Ns/m$ |
| Stiffness $K_p$           | $1.2 \times 10^8 N/m$    |                 |                          |

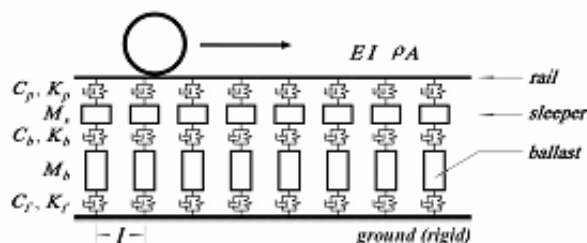


Figure 5 Example 2: A mass moving on a rail which is supported by the sleepers, ballast and spring and dashpot systems shown.

173 Figure 6 gives the PSD and variance of one static point at a support on the rail as the  
 174 mass passes it. It can be seen that, as might be expected, the responses are largest at high  
 175 load frequencies and when the moving mass is close to the point. The same conclusions are  
 176 drawn when the static point was taken midway between supports and the results are not shown  
 177 because they are very similar to Figure 6, e.g. the peak on Figure 6(b) was reduced by 10.66%.  
 178 Such examples could be extended without difficulty to allow for train wheels attached to bogies  
 179 moving on the track.

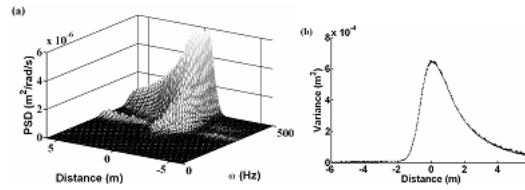


Figure 6 Vertical displacement responses of a static point at a support on the rail of Example 2. (a) PSD, (b) Variance.

## 180 5 CONCLUSIONS

181 Based on symplectic mathematics, a condensed equation of motion has been established for  
182 the loaded substructure of an infinitely periodic structure, the coefficient matrices of which  
183 are functions of the wave number. A Fourier expansion was then applied to separate the  
184 variables of time and wave number, which led to a recurrence scheme for computing the  
185 responses of the infinitely periodic structure. Finally, this method was combined with PEM to  
186 yield a convenient method for analyzing the random vibration of the structure. The proposed  
187 method was justified by a numerical example and was then applied to a more complicated  
188 time-dependent coupled system.

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